# **TECHNICAL NOTE**

# **Analysis of melting in an enclosure with discrete heating at constant rate**

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## **Introduction**

Cooling of electronic components has attracted many researchers in the heat transfer field. Natural-convection cooling by ambient air is quite popular because of its simplicity and reliability. Some published papers (Chu and Churchill 1976; Turner and Flack 1980) consider natural convection in an enclosure heated by discrete heat sources. Using solid-liquid phase change materials (PCM) to cool electronic components is a promising alternative for periodically operating apparatus, because phase-change cooling can provide relatively larger cooling capability than air. In accordance with this possibility, experiments were conducted by the present authors for investigating the melting process in an enclosure with one of its side walls discretely heated (Zhang et al. 1993).

Since the shape of the liquid phase is rather irregular during melting, and since the Rayleigh number may be very high (about  $10<sup>9</sup>$ ), a large amount of computer time is required for the numerical solution of the melting process (Ho and Viskanta 1984). Recently, boundary-layer theory has been adopted by some authors to solve the process of natural-convectiondominated melting. An analytical solution for the melting process in a rectangular enclosure isothermally heated from one of its vertical walls was obtained by Bejan (1989). A similar problem but with a boundary condition of uniform heat flux instead of uniform wall temperature was described by Zhang and Bejan (1989). But to the best of our knowledge, no analytical solution for the above-mentioned melting process with discrete heating has been published yet.

#### **Mathematical model**

The experimental enclosure used by Zhang et al. (1993) is schematically shown in Figure 1. Three discrete heat sources made of copper are flush-mounted on the lateral wall, which is made of acrylic Plexiglas. Heat leakage through the back of the lateral wall to the surroundings was minimized by guard heaters, and the whole enclosure was well insulated. The enclosure was filled with n-octadecane. During the beginning stage of the experiment, the surface temperatures of the sources increased gradually, but after a certain time period, they all reached some steady values and remained unchanged. This type of melting process is said to be quasi-steady. Once this state is established, all the heat supplied by the discrete sources will be

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used to melt the solid phase of PCM at the solid-liquid interface via the natural convection of the liquid phase.

In order to facilitate the solution process, the quasi-steady state is assumed in addition to some classical assumptions, e.g., the initial solid-phase temperature is isothermal and at the melting point. The solution procedure is basically the same as that employed by Bejan (1989), but an important improvement has been made by taking into account the conduction in the discretely heated wall. Since the heat leakage through the back surface of the wall has been minimized, the conduction in the wall can be simplified as a one-dimensional (I-D) steady-state conduction, the dimensionless heat diffusion equation is

$$
\frac{d}{dY}\left(K'_{\mathbf{w}}W\frac{d\theta_{\mathbf{w}}}{dY}\right) + Q^* = 0\tag{1}
$$

where  $K'_{\text{w}}$  is the dimensionless thermal conductivity of the vertical wall, which equals  $K<sub>h</sub>$  in the heated section or  $K<sub>w</sub>$  in



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the unheated section.  $Q^*$  is the difference between the amount of heat provided by the discrete sources and the amount transferred to the PCM. This can be expressed as

$$
Q^* = \begin{cases} \left(\frac{1}{3B} - \frac{2(\theta_w - \theta_c)}{\Lambda}\right) \text{Ra}^{1/5} & \text{for the heated section} \\ -\frac{2(\theta_w - \theta_c)}{\Lambda} \text{Ra}^{1/5} & \text{for the unheated section} \end{cases}
$$
(2)

The top and bottom of lateral wall are adiabatic. The boundary conditions, for Equation 1 are

$$
\frac{d\theta_{\mathbf{w}}}{dY} = 0, \text{ at } Y = 0 \quad \text{and} \quad 1
$$
 (3)

Mathematical formulations for the liquid region can be found in Zhang and Bejan's (1989) paper and are omitted here because of space limitations. A numerical method is used to solve these differential equations. The number of nodal points along the height is  $N = 900$  in the calculation. The convergence criterion used for both  $\theta_c$  and  $\theta_w$  is 10<sup>-9</sup>.

#### **Results and discussions**

The comparison between the predicted and measured dimensionless temperatures of the heated wall when  $Ste = 2.36$ is given in Figure 2. It can be seen that their general tendencies are about the same. Quantitatively speaking, the coincidence of the predicted and measured temperatures on the heat source surface is better than that on the unheated sections (i.e., the surface between the sources). This finding may be attributed to the 1-D simplification of the wall conduction. Compared to the results reported by Zhang and Bejan (1989), temperatures on the heat source surfaces are higher than those for uniform heating, and the reverse is true on the unheated sections. The average temperatures of the heated wall are about the same.

The Stefan number varied from 1.96 to 3.90 during the

#### **Notation**

- 
- $b$  Height of the heat source
- B Dimensionless height of the heat source, *b/H*
- $c_p$  Specific heat of liquid
- $\sigma$  Gravitational acceleration
- $H, L$  Height and width of enclosure
- $h_m$  Latent heat
- $k<sub>h</sub>$  Thermal conductivity of heated part of the wall
- $k_1$  Thermal conductivity of liquid  $k_{\rm w}$  Thermal conductivity of unheated part of the wall
- $K<sub>h</sub>$  Dimensionless thermal conductivity of heated
- part of the wall,  $k_h/k_1$  $K_{\rm w}$  Dimensionless thermal conductivity of unheated part of the wall,  $k_{w}/k_{1}$
- q Heat flux

 $g\beta(3qb/k<sub>l</sub>)H<sup>3</sup>$ Ra Rayleigh number,  $v\alpha$ 

$$
Ste \tStefan number, \frac{c_p(3qb/k_l)}{2}
$$

- $h_{\mathbf{m}}$
- $T_{\rm m}$  Core temperature<br>  $T_{\rm m}$  Melting point<br>  $T$  Wall temperature
- Melting point

$$
T_w
$$
 Wall temperature  
 $U_0$  Dimensionless local interface velocity

$$
U_0
$$
 Dimensions local interface vector



Figure 2 Dimensionless wall temperature

experiment conducted by Zhang et al. (1993). The computational results show that the dimensionless temperatures of the sources are virtually independent of Ste. So for Ste  $= 1.96 - 3.90$ , the surface temperatures of heat sources  $T<sub>h</sub>$  can be calculated as

$$
(T_h)_{n} - T_m = A_n \frac{3qb}{k} \text{Ra}^{-1/5}
$$
 (4)

where  $n = 1, 2, 3$  represents the lower, middle, and upper heat sources respectively, and  $A_n$ 's denote corresponding constants, which in that experiment are

$$
A_1 = 1.67 \qquad A_2 = 2.41 \qquad A_3 = 3.32
$$

Thickness of discrete sources W W Dimensionless thickness of discrete sources, *w/H*  Vertical coordinate Y Y Dimensionless vertical coordinate, *y/H Greek symbols*  Thermal diffusivity of liquid  $\pmb{\alpha}$  $\beta$  Volumetric thermal expansion coefficient of liquid  $\delta$  Thickness of cold boundary layer  $\Delta$  Dimensionless cold boundary-layer thickness,  $\frac{\delta}{\epsilon}$  Ra<sup>1/5</sup> H  $\theta_{\rm c}$  Dimensionless core temperature,  $(T_c - T_m)$  Ra<sup>1/5</sup>  $3qb/k<sub>1</sub>$  $\theta_{\rm w}$  Dimensionless wall temperature,  $\frac{(T_w - T_m)}{(T_w - T_m)}$  Ra<sup>1/5</sup>  $3qb/k<sub>1</sub>$  $\lambda$  Thickness of warm boundary layer A Dimensionless warm boundary-layer thickness,  $\lambda$  $Ra^{1/5}$ H v Kinematic viscosity of liquid  $\rho$  Density of liquid



*Figure 3* Effect of dimensions of heat sources on the temperature profile

The effect of dimensions of the heat source on the temperature profile in the enclosure is illustrated in Figure 3. As to the effect of the source dimensions, under the same Stefan number (i.e., the power supplied to the discrete source remains unchanged), the larger the height  $b$ , the lower the source surface temperature will be. However, the temperature of the unheated section increases. In other words, the temperature difference between the heated and unheated sections will decrease, and then the longitudinal conduction of the wall will decrease as the height of the source increases. The variation of the dimensionless solid-liquid interface velocity along the height of the enclosure with different source heights is shown in Figure 4. This dimensionless velocity can be calculated from the energy conservation at the solid-liquid interface (Bejan 1989), and the result is  $U_0(Y) = 2\theta_e(Y)/\Delta(Y)$ . From Figure 4, the source height has only a little effect on the interface velocity. The shape of



*Figure 4* Effect of dimensions of heat sources on the solid-liquid interface velocity



*Figure 5* Effect of unheated wall conductivity on the temperature profile

solid-liquid interface becomes more smooth with increasing source height because the temperature difference between the heated section and unheated section becomes smaller.

The variations of wall temperature for different unheated wall conductivities are shown in Figure 5. The surface temperature of the upper source decreases and that of the lower one increases with increasing unheated wall conductivity. The temperatures of the unheated sections are increased as well. The surface temperature distribution of the whole wall tends to be uniform, because the amount of heat flowing downwards through the wall increases with the increasing unheated wall thermal conductivity. Figure 6 shows the variations of the interface moving velocity when the unheated wall conductivity varies. The velocity of the lower interface increases and that of the upper interface decreases as the unheated wall conductivity increases. The reason for this variation is that when unheated



*Figure 6* Effect of unheated wall conductivity on the solid-liquid interface velocity

wall conductivity increases, the temperature of the upper part of the wall decreases, but the temperature of the lower part increases.

## **Acknowledgment**

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